

Thermal Materials Workshop 2001

Application of Foam Metal Technology to Aircraft Systems- Direction and Status

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NORTHROP GRUMMAN

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Program Participants

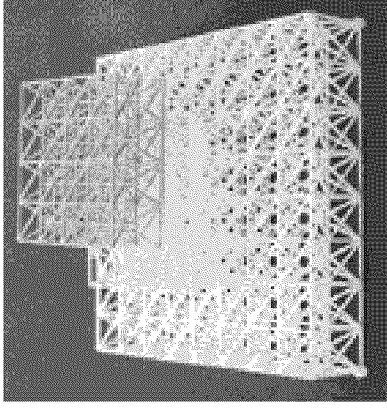
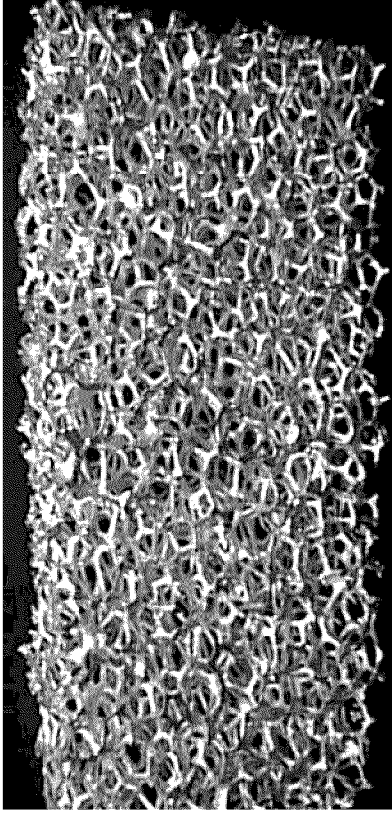
- Northrop-Grumman Corporation
 - Technology Development, AEW&EW systems business area
 - Logicon
- ONR / DARPA
- Ozer Engineering
- ERG

Introduction

- **Outgrowth of DARPA Ultra-Lightweight Metallic Structures Program**
 - **GASAR Component Design, Production, Test, and Cost-benefit Study (our introduction to this community)**
 - **MURI (foams and periodic structures)**
- **Present Project**
 - **Structurally Integrated Thermal Management of Airborne Early Warning & Electronic Warfare Systems**
 - **Technology Transition**
 - **DARPA Synthetic Multi-Functional Materials**
 - **Rules and Tools, Relevant Database**
 - **E-2C, Other Applications**

Technical Areas

- Generic Technologies - Thermal and Structural Behavior of Metal Foams and Lattice Structures



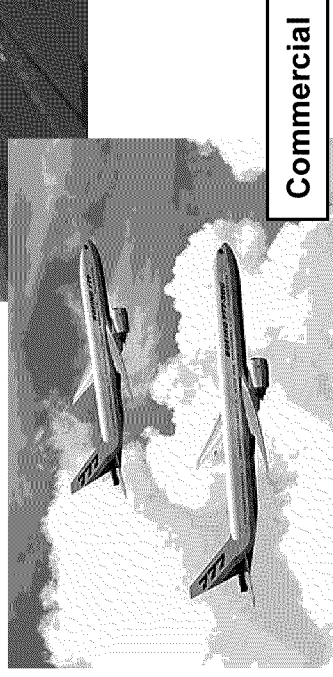
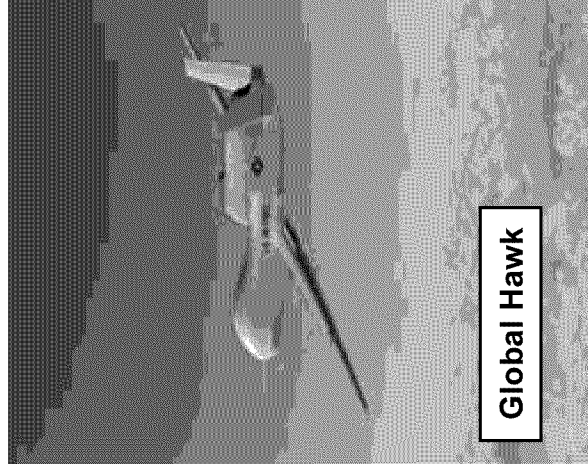
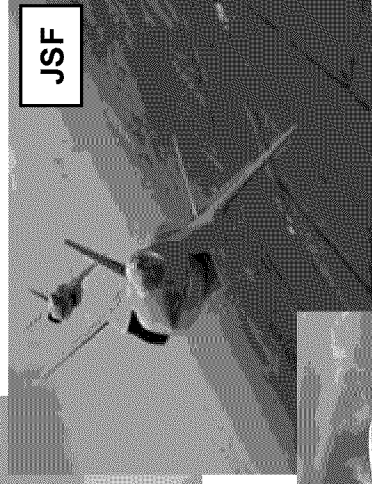
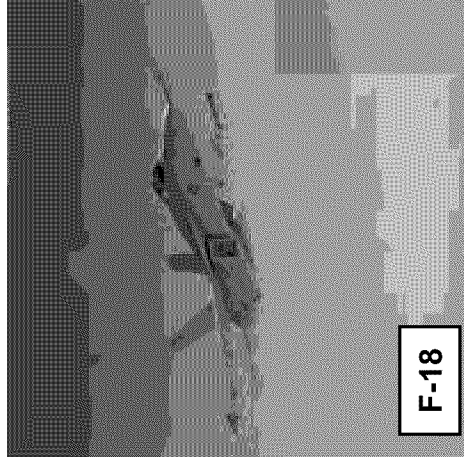
- Specific Applications - Thermal Management for Airborne Early Warning & Electronic Warfare Systems



Applications

- E-2C Heat Exchanger Cores
- E-2C Avionics Racks
- EA-6B, F/A-18E/F, F/A-18G, JSF
- Unmanned Air Vehicles
- Commercial

High Efficiency Porous Metal Heat Exchangers



Project Tasks

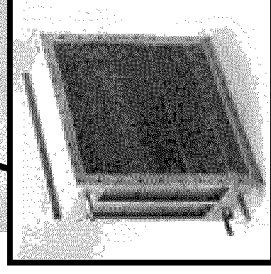
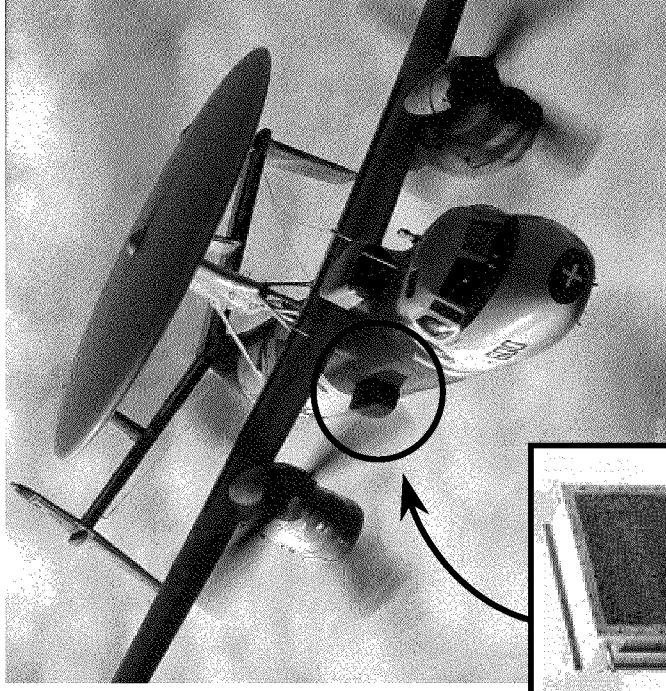
- **1a. E-2C LCS Heat Exchanger Element - Design, Fabrication and Test**
- **2a. Heat Exchanger Elements for E-2C Avionics Racks – Concepts**

E-2C Air/Liquid Heat Exchanger

- Conventional Redesign of Air/Liquid Heat Exchanger to Accommodate Increased Load Plus Growth Large and Heavy
- Porous Metal Heat Exchanger Offers the Potential to be:

- Lighter
- Smaller
- Less Expensive

Than Conventional Redesign

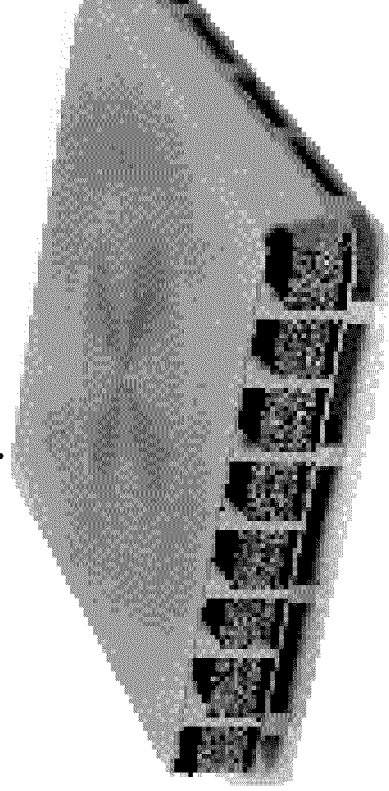


Foam Metal
Heat Exchanger



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Integrated Systems
Sector

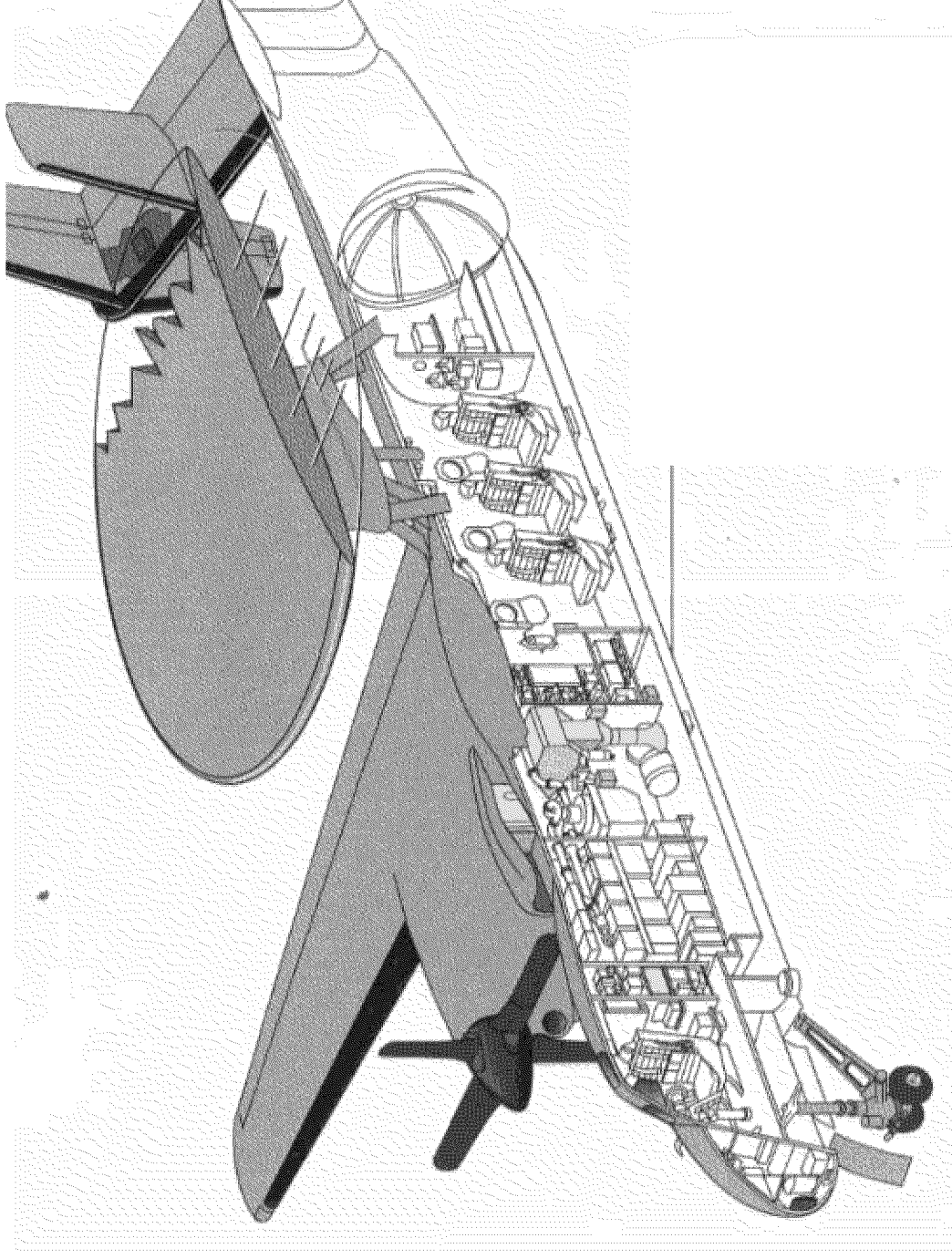
E-2C Integrated Avionics Rack



Integrated Avionics Rack

- New Avionics Suite and Redesign of Racks and Cards and Allow for an Integrated Approach to System Cooling
- Opportunity to Address Thermal Management by Incorporating Integrated Cooling Capability
- Integrated Racks Offer Weight, Performance, Size Advantages

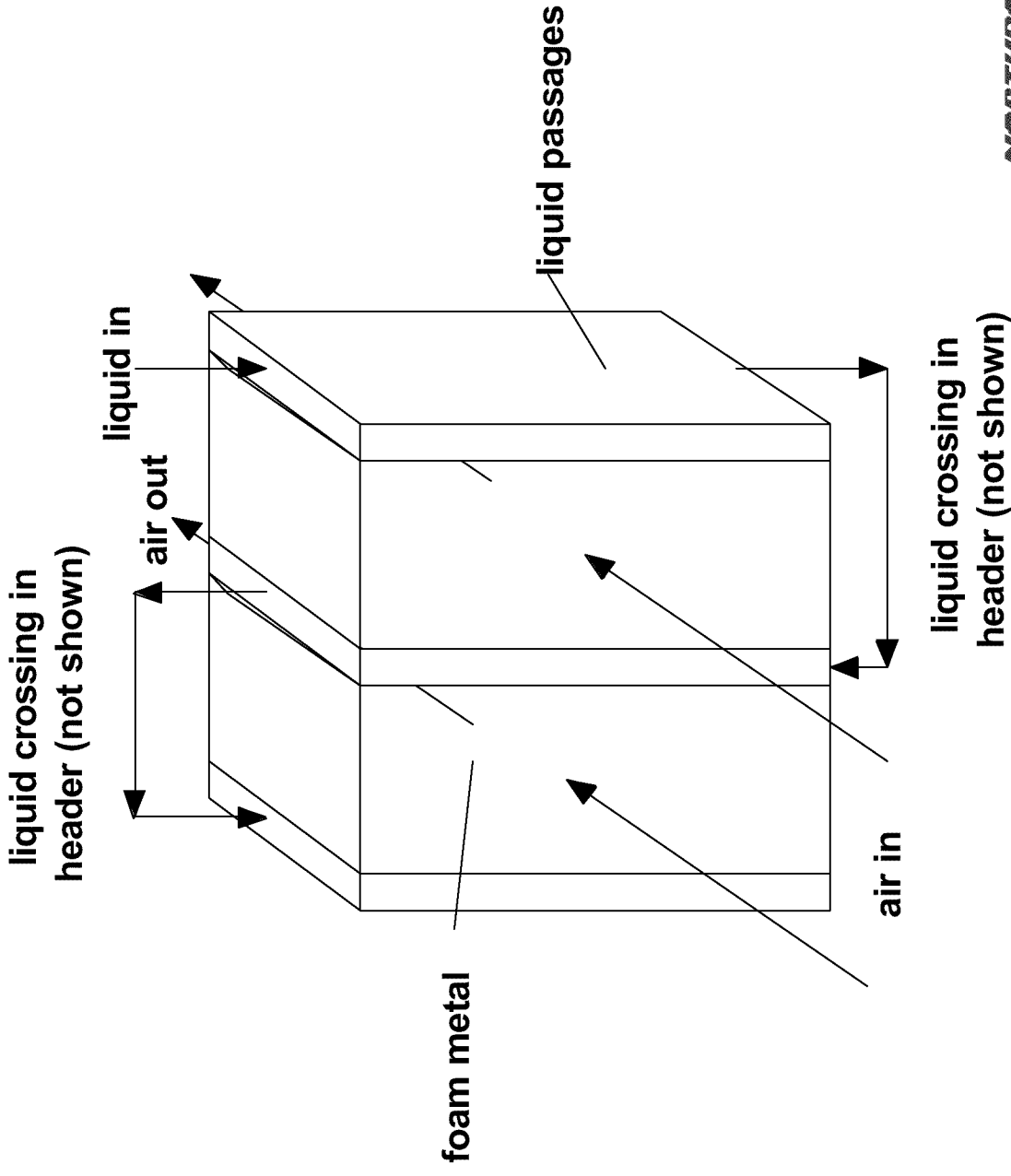
E-2C Interior



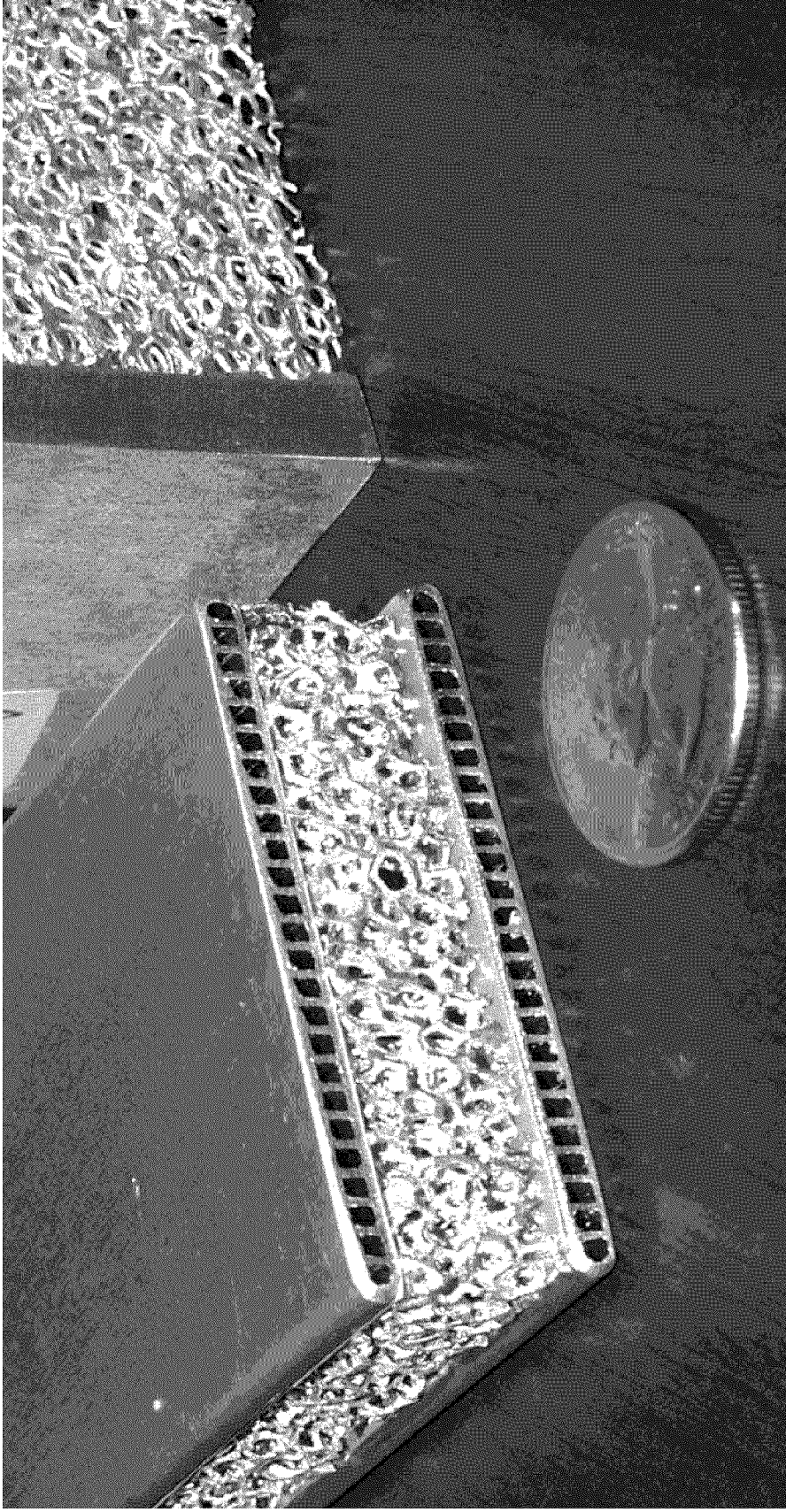
Structurally Integrated Thermal Management of Airborne
Early Warning & Electronic Warfare Systems - 3

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Integrated Systems
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Foam Metal Heat Exchanger Basic Concept



Manufacturing Demo Samples (ERG)

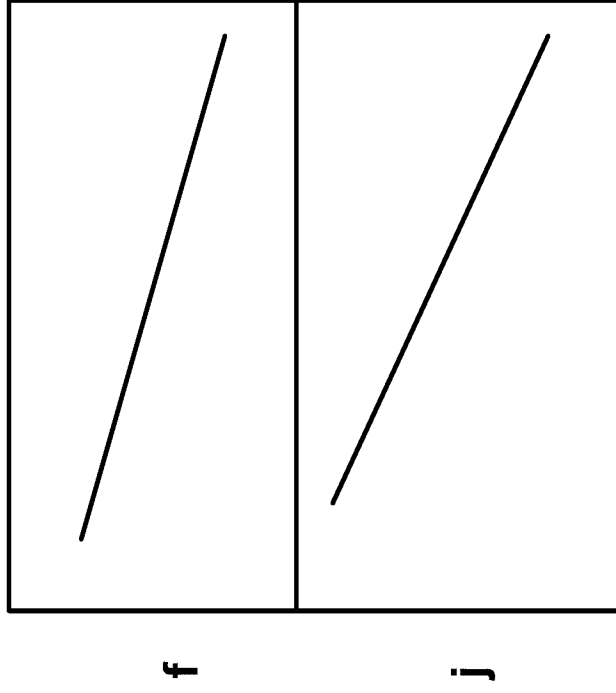


Structurally Integrated Thermal Management of Airborne
Early Warning & Electronic Warfare Systems - 2

Data requirements

- Heat transfer and pressure drop vs Reynolds Number
 - Compressed Foams
 - h_{eff} , ie $\eta h_{true} A_{es}$
 - $\Delta P/L$
 - Present In preferred Kays & London (K&L) Heat Transfer Data Format
 - Ref: Compact Heat Exchangers, Kays & London

Kays & London Data Presentation



Friction factor

$$f = \frac{\Delta p}{4 \left(\frac{L}{D_h} \right) \left(\frac{\rho V^2}{2} \right)}$$

Colburn j factor

$$j = \frac{h}{\rho c_p V} Pr^{2/3}$$

$$= St * Pr^{2/3}$$

Definition of Terms

- Flow velocity
 - \dot{V} = mass flow rate / (ρA_{\min}) , where A_{\min} is the minimum flow area.
 - matrices $G = \rho \dot{V} = W / (p A_{fr})$
- Hydraulic diameter:
- $D_h = 4 \times (\text{minimum flow area}) / (\text{total heat transfer area})$
- L = flow length of heat exchanger
- $R_e = \rho \dot{V} D_h / \mu$

Projected Heff 10 ppi foam-sea level

NORMALIZED HEAT FLUX FROM THE BASE

10 PPI OPEN CELL ALUMINUM FOAM

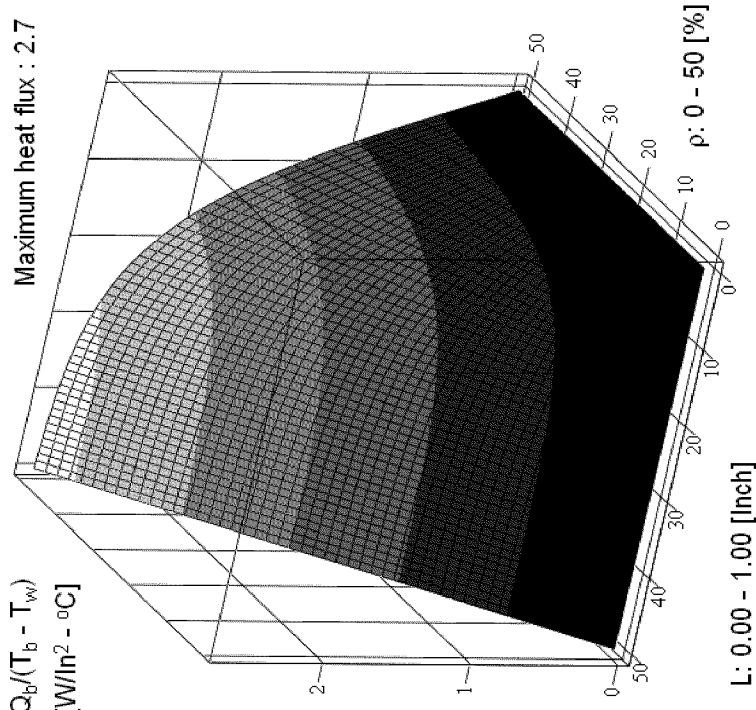
COOLANT : AIR

Foam thickness (L), Relative density (ρ)

Preliminary

$Q_b / (T_b - T_w)$
[W/in² - °C]

Maximum heat flux : 2.7



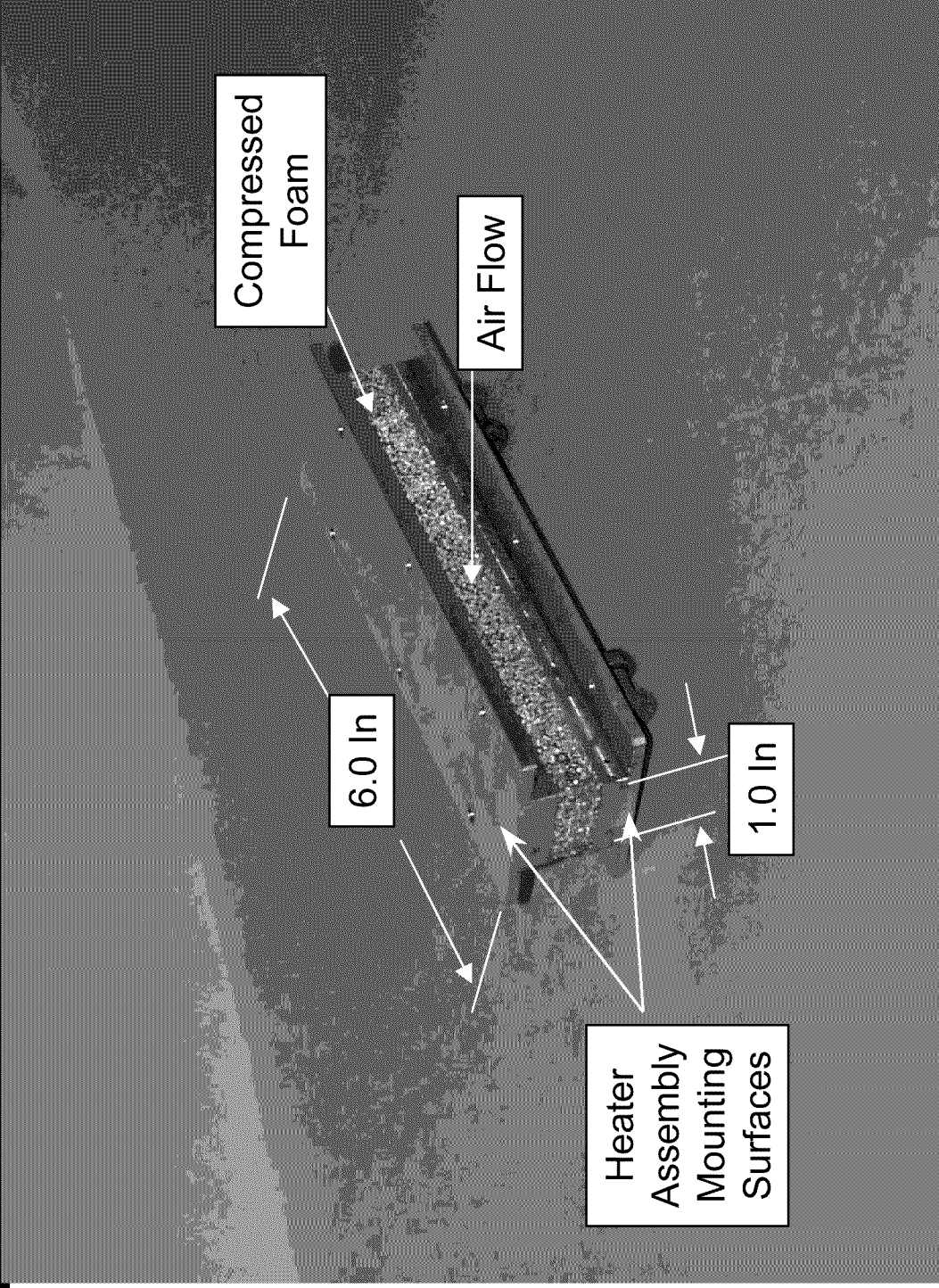
Heat Exchanger Design Methodology

- Kays & London type measured data for extended surface
 - pressure drop vs. Re
 - heat transfer vs. Re
- Calculate h using K&L type data
- Apply same h to both extended surface and wall
- Calculate extended surface temperature effectiveness based on h , mat'l and geometry
- Calculate effective total heat exchange area

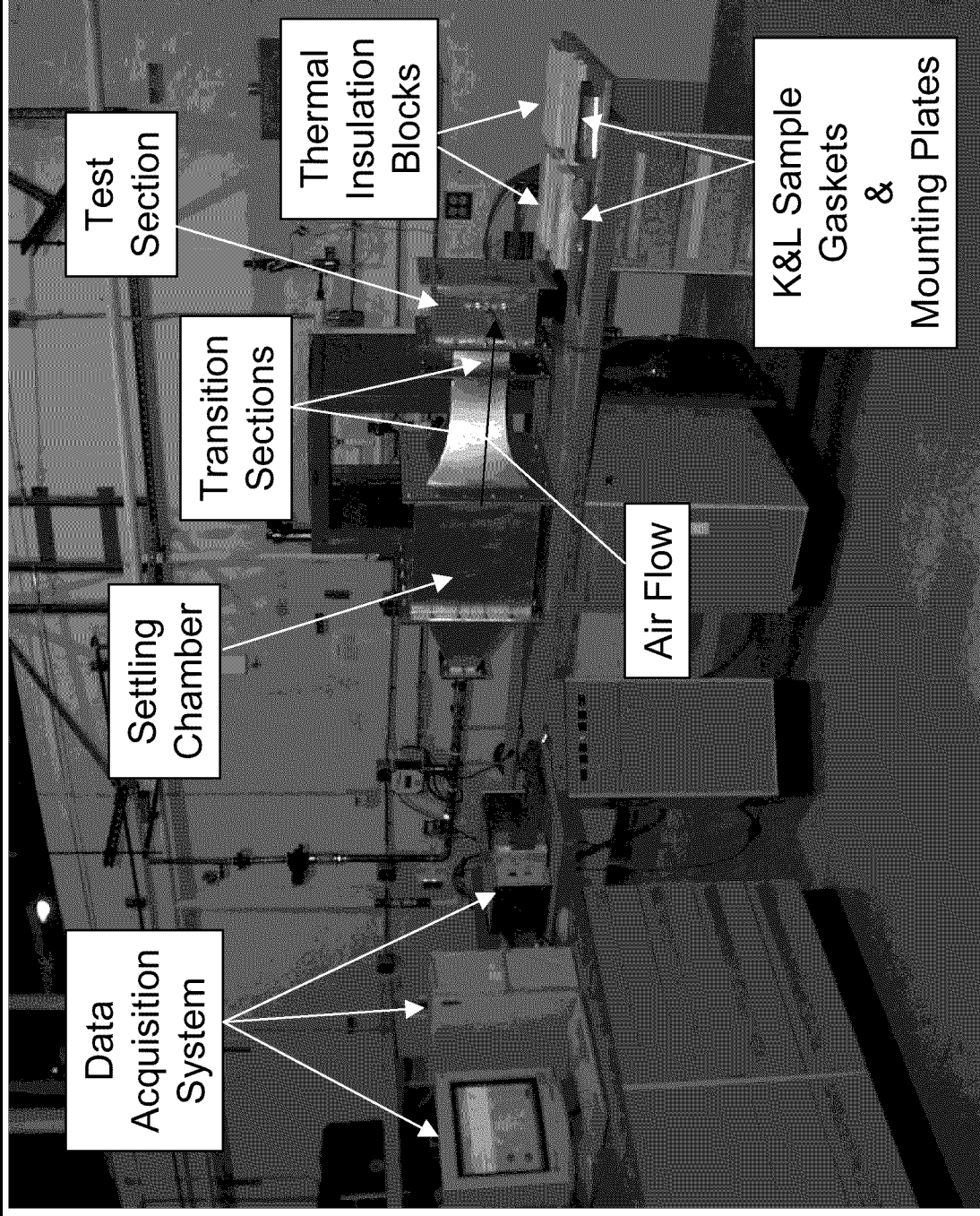


K&L Test Apparatus

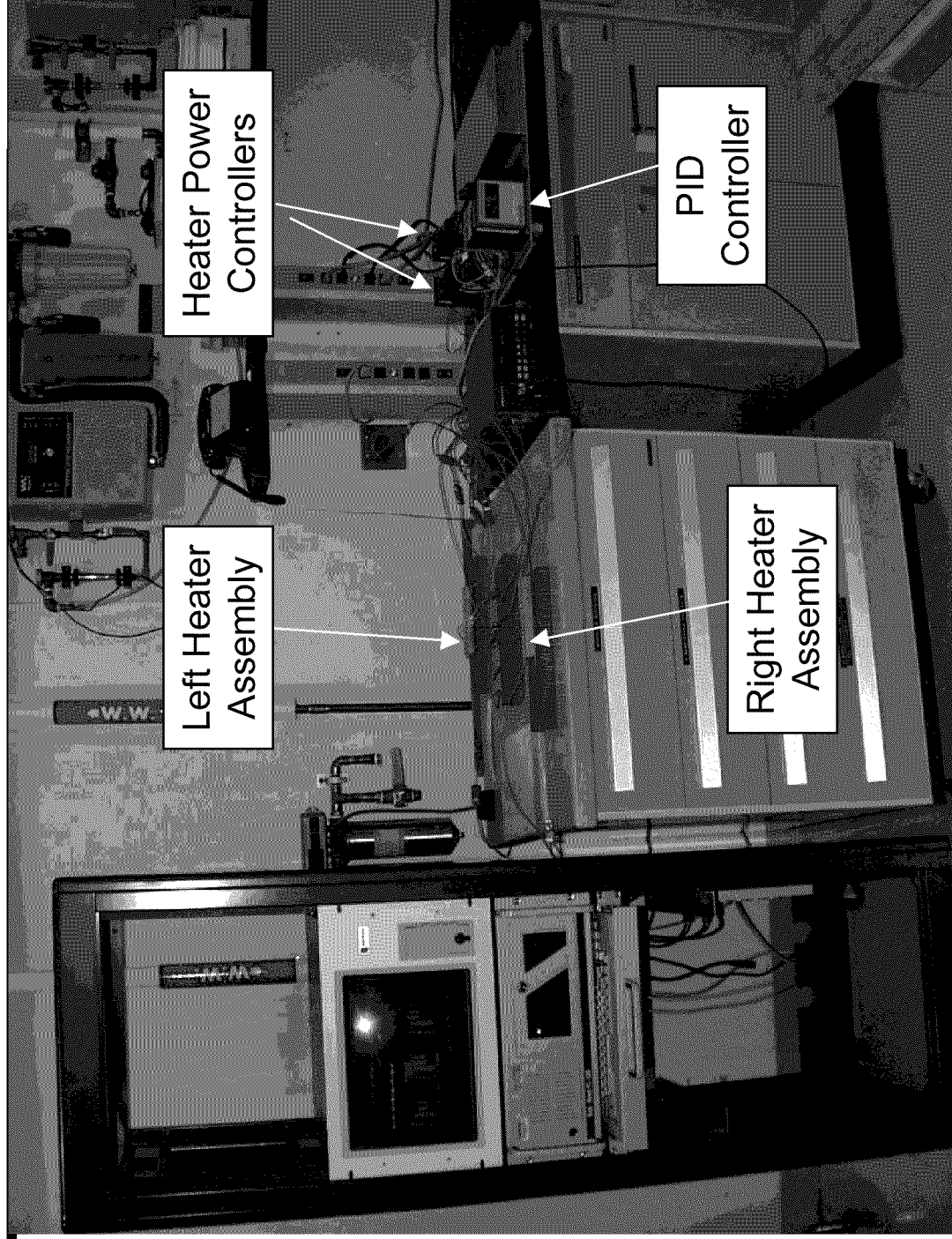
NGC KAYS & LONDON TEST SAMPLE



KAYS & LONDON TEST APPARATUS



K&L HEATER POWER PID CONTROL TEST



Summary - Technical Status / Direction

- **E-2C LCS heat exchanger**
 - **Completed**
 - **LCS HX conceptual design**
 - **LCS HX initial sizing**
 - **K&L test apparatus design, fabrication, checkout**
 - **data acquisition system**
 - **temperature controller testing**
 - **air flow supply checkout**
 - **Instrument calibration**
 - **NGC K&L testing start 1-2 weeks**
 - **Foam metal HX sub element fabrication demo**
 - **Design system integration methodology**

Summary - Technical Status / Direction (cont.)

- E-2C LCS work in progress
 - K&L sample testing
 - HX element fabrication demonstration
 - HX element thermal performance demonstration
 - Design system integration method verification
- E2C Integrated avionics racks
 - preliminary concepts
- Structurally integrated heat exchangers
 - preliminary concepts
- Additional specific aircraft applications